

CE 415

DESIGN OF STEEL STRUCTURES

LECTURE 13

COMPRESSION MEMBER

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OUTLINE

- Local buckling
- Column Biaxial bending
- Biaxial bending math problem

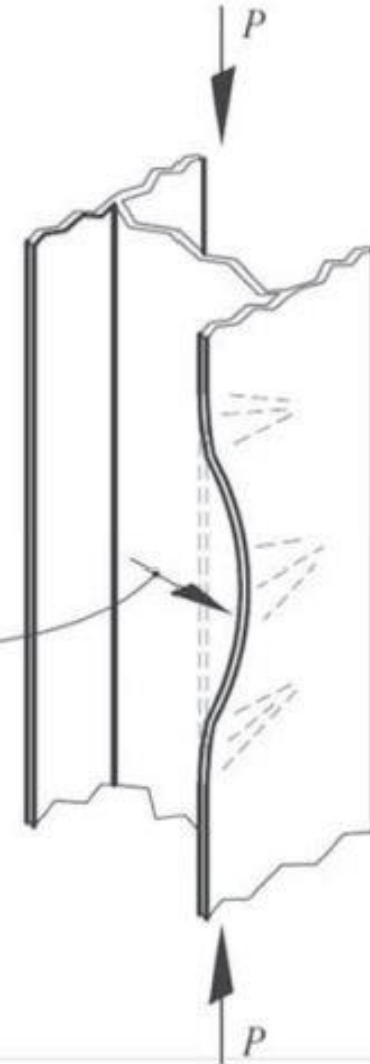
LOCAL BUCKLING

Local buckling is a phenomenon by which a portion of the section of a column or beam buckles instead of overall buckling.

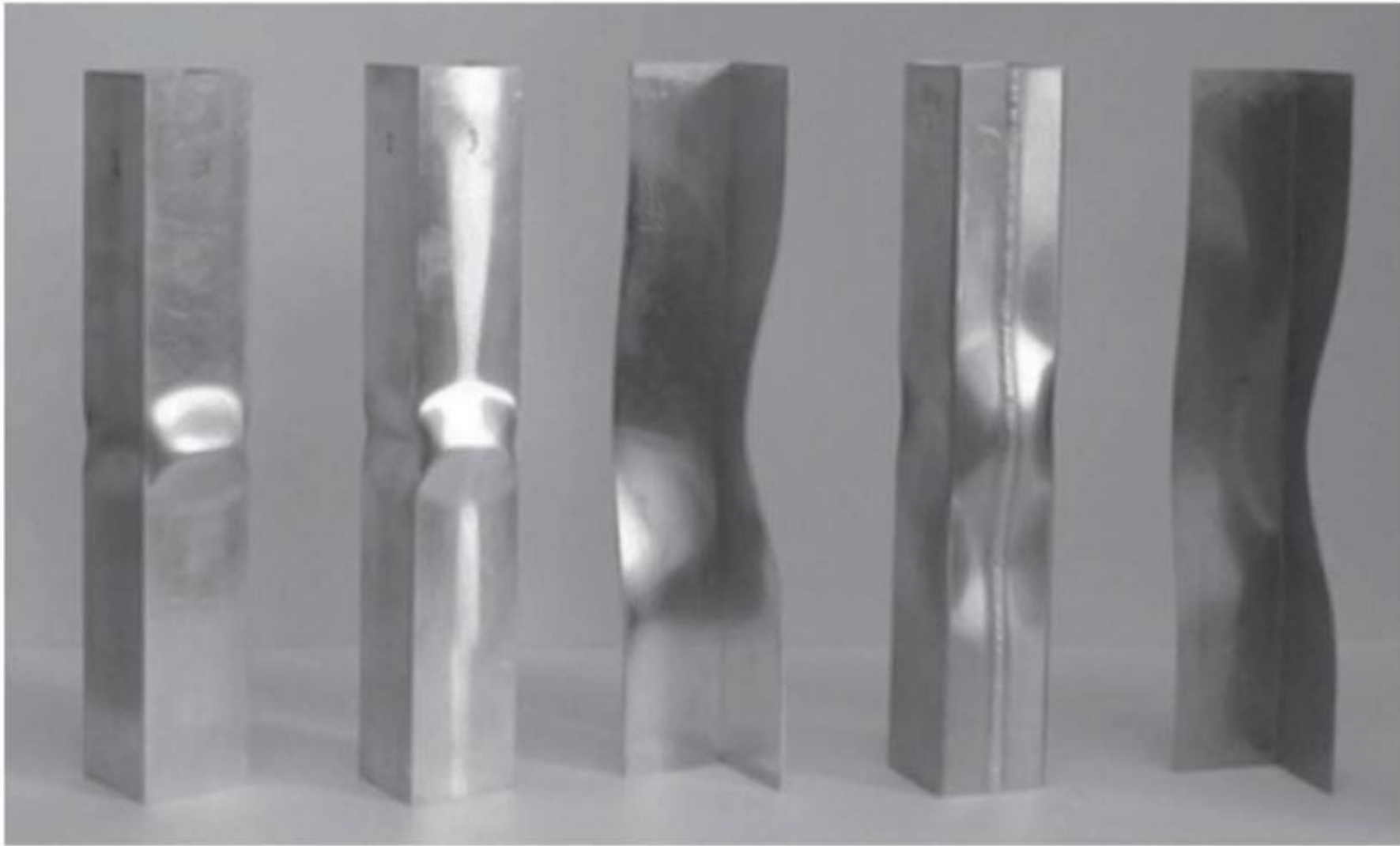
Local buckling leads to a reduction in the strength of a compression member and prevents the member from reaching its overall compression capacity.

To avoid or prevent local buckling, the AISC specification prescribes limits to the width-to-thickness ratios of the plate components that make up the structural member.

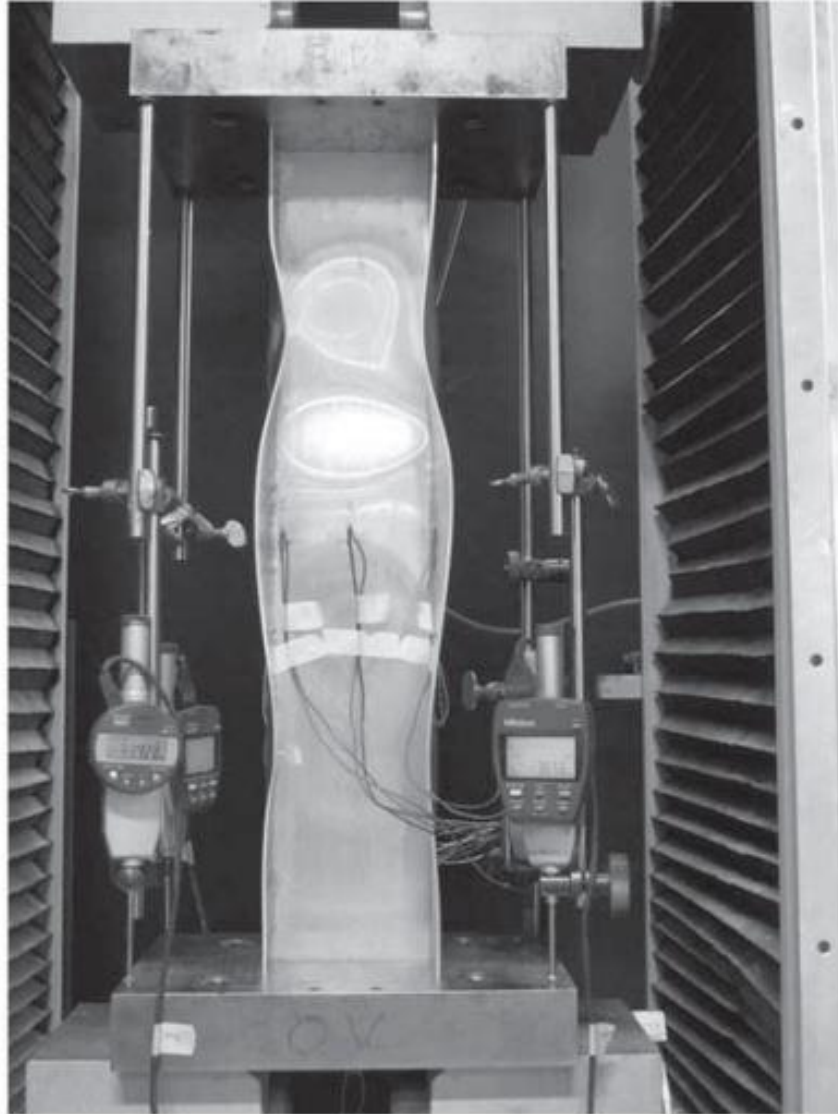
localized buckling of
column flange under
compression stress



LOCAL BUCKLING



LOCAL BUCKLING



Local buckling primarily depends on the ratio, b/t , of the width (b) and thickness (t) of the plate elements that builds up a section.

LOCAL BUCKLING

Based on the width/thickness ratio steel sections are defined as

- Compact:** A compact section reaches its cross-sectional material strength, or capacity, before local buckling occurs.
- Non-Compact:** In a non-compact section, only a portion of the cross-section reaches its yield strength before local buckling occurs.
- Slender:** In a slender section, the cross-section does not yield and the strength of the member is governed by local buckling.

The use of slender sections as compression members is not efficient or economical; therefore, the use of slender section in design practice is not recommended.

Designation	Nom. Weight, lb/ft	Area, A, in ²	Depth, d, in	Flange		Web Thickness, t _w , in	Axis X-X			Axis Y-Y		
				Width, b _f , in	Thickness, t _f , in		I, in ⁴	S, in ³	r, in	I, in ⁴	S, in ³	r, in
W10 x 68	68	20	10.4	10.13	0.77	0.47	394	75.7	4.44	134	26.4	2.59
W10 x 60	60	17.6	10.22	10.08	0.68	0.42	341	66.7	4.39	116	23	2.57
W10 x 54	54	15.8	10.09	10.03	0.615	0.37	303	60	4.37	103	20.6	2.56
W10 x 49	49	14.4	9.98	10	0.56	0.34	272	54.6	4.35	93.4	18.7	2.54
W10 x 45	45	13.3	10.1	8.02	0.62	0.35	248	49.1	4.32	53.4	13.3	2.01
W10 x 39	39	11.5	9.92	7.985	0.53	0.315	209	42.1	4.27	45	11.3	1.98
W10 x 33	33	9.71	9.73	7.96	0.435	0.29	170	35	4.19	36.6	9.2	1.94
W10 x 30	30	8.84	10.47	5.81	0.51	0.3	170	32.4	4.38	16.7	5.75	1.37
W10 x 26	26	7.61	10.33	5.77	0.44	0.26	144	27.9	4.35	14.1	4.89	1.36
W10 x 22	22	6.49	10.17	5.75	0.36	0.24	118	23.2	4.27	11.4	3.97	1.33

EXAMPLE:

Determine the allowable compressive load carrying capacity of the column shown in Fig. It consists of W10×45 section having A992 ($F_y = 50$ ksi) steel. There are hinge support at top and bottom that allows rotation in any direction. Also the column has weak direction support (braced) at mid-height so that lateral deflection is prevented in x direction. Use ASD approach.

SOLUTION:

For W10×45 section, from AISC Manual Chart we have $A = 13.3 \text{ in}^2$, $r_x = 4.32 \text{ in.}$, $r_y = 2.01 \text{ in.}$

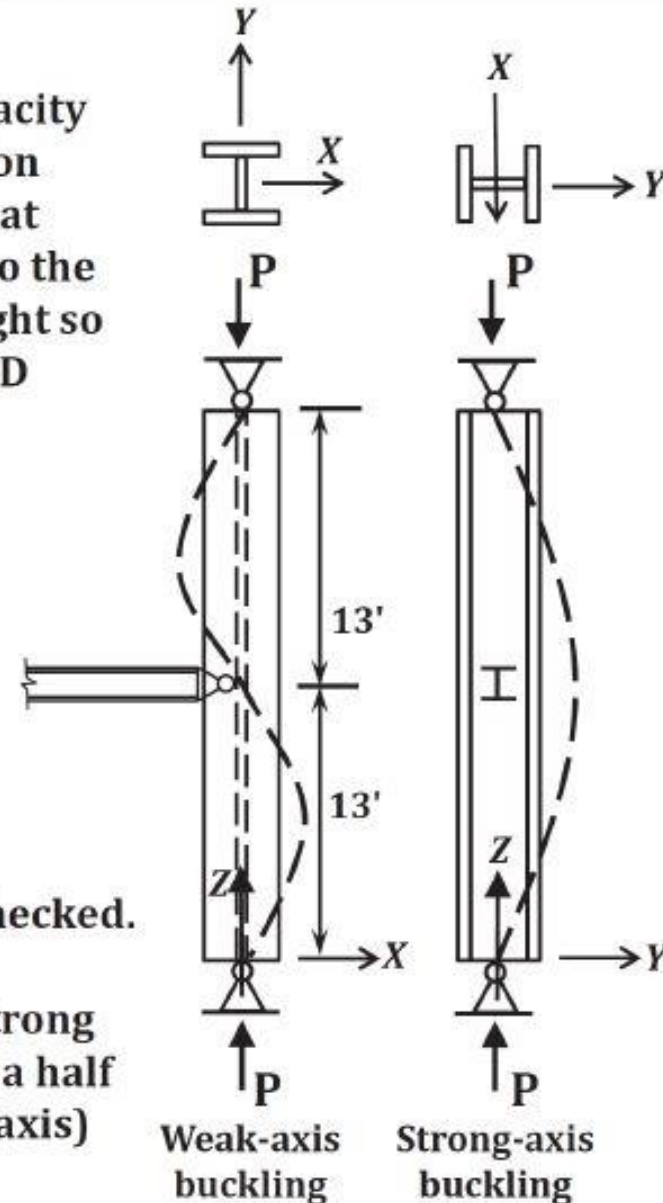
$x \rightarrow$ strong axis

$y \rightarrow$ weak axis

Column length, $L = (13 \times 2) \times 12 = 312 \text{ in.}$

Possibility of buckling in both x and y directions to be checked.

Buckling in y direction causes bending about x axis or strong axis. For strong axis buckling, the buckling shape is like a half sine wave over full column length. Thus for strong (or x axis) axis buckling, $K_x = 1.0$



$$\therefore K_x L / r_x = 1.0 \times 312 / 4.32 = 72.22$$

$$\therefore F_{ex} = \pi^2 E / (K_x L / r_x)^2 = 3.14^2 \times 29000 / (72.2)^2 = 54.82 \text{ ksi. } (> F_y, \text{ note})$$

$$\text{And } 4.71 \sqrt{E / F_y} = 4.71 \sqrt{(29000 / 50)} = 113.43 \therefore K_x L / r_x < 4.71 \sqrt{E / F_y}$$

$$F_{cr} = \left[0.658 \frac{F_y}{F_{ex}} \right] F_y = [0.658^{(50/54.82)}] 50 = 34.13 \text{ ksi}$$

Nominal strength for x-axis buckling $P_{nx} = F_{cr} A_g = 34.13 \times 13.3 = 454 \text{ kip}$

Buckling in x direction causes bending about y axis or weak axis. For weak axis buckling, the buckling shape is like a full sine wave over full column length. Thus for weak (or y axis) axis buckling, $K_y = 0.5$

$$\therefore K_y L / r_y = 0.5 \times 312 / 2.01 = 77.61$$

$$\therefore F_{ey} = \pi^2 E / (K_y L / r_y)^2 = 3.14^2 \times 29000 / (77.61)^2 = 47.47 \text{ ksi.}$$

$$F_{cr} = \left[0.658 \frac{F_y}{F_{ey}} \right] F_y = [0.658^{(50/47.47)}] 50 = 32.17 \text{ ksi}$$

Nominal strength for y axis buckling $P_{ny} = F_{cr} A_g = 32.17 \times 13.3 = 427.9 \text{ kip}$

$\therefore P_n = \text{smaller of } P_{nx} \text{ and } P_{ny} = 427.9 \text{ kip}$

$\therefore \text{Allowable strength } P = P_n / \Omega = 427.9 / 1.67 = 256.2 \text{ kip}$