

# CE 415 DESIGN OF STEEL STRUCTURES

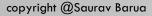
LECTURE 13
COMPRESSION MEMBER

SEMESTER: SUMMER 2021

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## **OUTLINE**

- Local buckling
- Column Biaxial bending
- Biaxial bending math problem

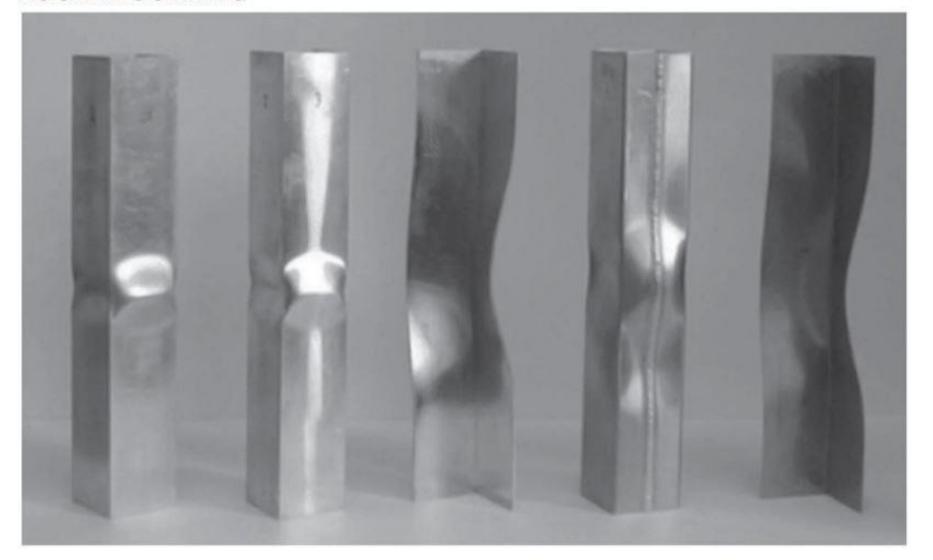
Local buckling is a phenomenon by which a portion of the section of a column or beam buckles instead of overall buckling.

Local buckling leads to a reduction in the strength of a compression member and prevents the member from reaching its overall compression capacity.

To avoid or prevent local buckling, the AISC specification prescribes limits to the width-to-thickness ratios of the plate components that make up the structural member.

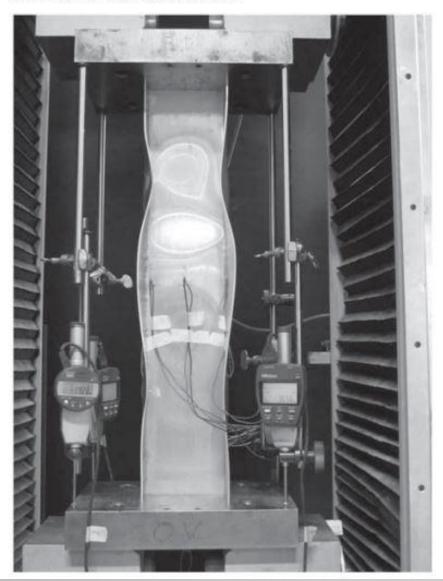
localized buckling of column flange under compression stress





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Local buckling primarily depends on the ratio, b/t, of the width (b) and thickness (t) of the plate elements that builds up a section.

Based on the width/thickness ratio steel sections are defined as

**Compact:** A compact section reaches its cross-sectional

material strength, or capacity, before local

buckling occurs.

Non-Compact: In a non-compact section, only a portion of

the cross-section reaches its yield strength before

local buckling occurs.

Slender: In a slender section, the cross-section does not

yield and the strength of the member is governed

by local buckling.

The use of slender sections as compression members is not efficient or economical; therefore, the use of slender section in design practice is not recommended.

Desig- nation	Nom. Weight, Ib/ft	Area, A, in2	Depth, d, in	Flange		Web Thickness	Axis X-X			Axis Y-Y		
				Width, br, in	Thick- ness, ttr, in	t <sub>w</sub> , in	I, in4	S, ins	r, in	I, in4	S, ins	r, in
W10 x 68	68	20	10.4	10.13	0.77	0.47	394	75.7	4.44	134	26.4	2.59
W10 x 60	60	17.6	10.22	10.08	0.68	0.42	341	66.7	4.39	116	23	2.57
W10 x 54	54	15.8	10.09	10.03	0.615	0.37	303	60	4.37	103	20.6	2.56
W10 x 49	49	14.4	9.98	10	0.56	0.34	272	54.6	4.35	93.4	18.7	2.54
W10 x 45	45	13.3	10.1	8.02	0.62	0.35	248	49.1	4.32	53.4	13.3	2.01
W10 x 39	39	11.5	9.92	7.985	0.53	0.315	209	42.1	4.27	45	11.3	1.98
W10 x 33	33	9.71	9.73	7.96	0.435	0.29	170	35	4.19	36.6	9.2	1.94
W10 x 30	30	8.84	10.47	5.81	0.51	0.3	170	32.4	4.38	16.7	5.75	1.37
W10 x 26	26	7.61	10.33	5,77	0.44	0.26	144	27.9	4.35	14.1	4.89	1.36
W10 x 22	22	6.49	10.17	5.75	0.36	0.24	118	23.2	4.27	11.4	3.97	1.33



Determine the allowable compressive load carrying capacity of the column shown in Fig. It consists of W10×45 section having A992 ( $F_y$  = 50 ksi) steel. There are hinge support at top and bottom that allows rotation in any direction. Also the column has weak direction support (braced) at mid-height so that lateral deflection is prevented in x direction. Use ASD approach.

#### SOLUTION:

For W10×45 section, from AISC Manual Chart we have  $A = 13.3 \text{ in}^2$ ,  $r_x = 4.32 \text{ in.}$ ,  $r_v = 2.01 \text{ in.}$ 

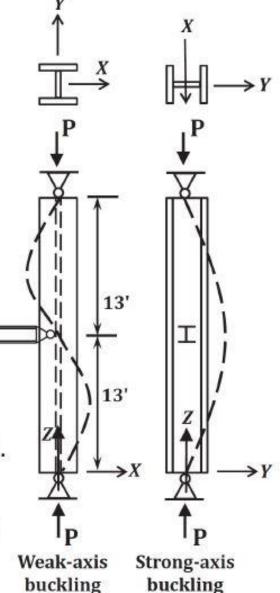
 $x \rightarrow$  strong axis

 $y \rightarrow$  weak axis

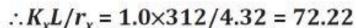
Column length,  $L = (13 \times 2) \times 12 = 312$  in.

Possibility of buckling in both x and y directions to be checked.

Buckling in y direction causes bending about x axis or strong axis. For strong axis buckling, the buckling shape is like a half sine wave over full column length. Thus for strong (or x axis) axis buckling,  $K_x = 1.0$ 



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$$\therefore F_{ex} = \pi^2 E/(K_x L/r_x)^2 = 3.14^2 \times 29000/(72.2)^2 = 54.82 \text{ ksi. (>} F_y, \text{ note)}$$
And  $4.71\sqrt{(E/F_y)} = 4.71\sqrt{(29000/50)} = 113.43 \therefore K_x L/r_x < 4.71\sqrt{(E/F_y)}$ 

$$F_{\rm cr} = \left[0.658^{\frac{F_y}{F_e}}\right] F_y = [0.658^{(50/54.82)}] 50 = 34.13 \text{ ksi}$$

Nominal strength for x-axis buckling  $P_{nx} = F_{cr} A_g = 34.13 \times 13.3 = 454$  kip

Buckling in x direction causes bending about y axis or weak axis. For weak axis buckling, the buckling shape is like a full sine wave over full column length. Thus for weak (or y axis) axis buckling,  $K_y = 0.5$ 

$$K_yL/r_y = 0.5 \times 312/2.01 = 77.61$$

$$\therefore F_{ey} = \pi^2 E / (K_y L / r_y)^2 = 3.14^2 \times 29000 / (77.61)^2 = 47.47 \text{ ksi.}$$

$$F_{\rm cr} = \left[0.658^{\frac{F_y}{F_e}}\right] F_y = [0.658^{(50/47.47)}] 50 = 32.17 \text{ ksi}$$

Nominal strength for y axis buckling  $P_{ny} = F_{cr} A_g = 32.17 \times 13.3 = 427.9$  kip

$$\therefore P_n = \text{smaller of } P_{nx} \text{ and } P_{ny} = 427.9 \text{ kip}$$

:. Allowable strength  $P = P_n/\Omega = 427.9/1.67 = 256.2$  kip

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