



# CE 415

# DESIGN OF STEEL STRUCTURES

LECTURE 11  
MIDTERM REVIEW

SEMESTER: SUMMER 2020

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# OUTLINE

- MIDTERM SYLLABUS

LECTURE 1 (16.5.20) – LECTURE 7 (13.6.20)

MARKS                  25 MARKS

- 2 questions with a (Math), b and c (theory type) segments
- Time 2hr
- No alternatives options

# LRFD & ASD

## Load and Resistance Factor Design (LRFD)

Resistance or strength:  $R_n$

Resistance factor:  $\phi < 1.0$

Load effect:  $Q_i$  [  $i = 1, 2, 3, \dots$  for dead load, live load etc.]

Load factor:  $\gamma_i$  [  $i = 1, 2, 3, \dots$  for dead load, live load etc.]

$R_u$ : Factored load effect

LRFD safety requirement

$$\phi R_n \geq R_u (= \sum \gamma_i Q_i)$$

## Allowable Strength Design (ASD)

Resistance or strength:  $R_n$

Safety factor:  $\Omega > 1.0$

Load effect:  $Q_i$  [  $i = 1, 2, 3, \dots$  for dead load, live load etc.]

Load factor:  $\gamma_i = 1.0$  [  $i = 1, 2, 3, \dots$  for dead load, live load etc.]

$R_a$ : Required strength / allowable strength

Saurav Barua

- The two distinct procedures employed by designers are,
  - Allowable Strength Design (ASD) &
  - Load & Resistance Factor Design (LRFD).

**Effective hole diameter**      = Actual hole diameter +  $\frac{1}{16}$ "  
                                        = Bolt diameter +  $\frac{1}{8}$ "

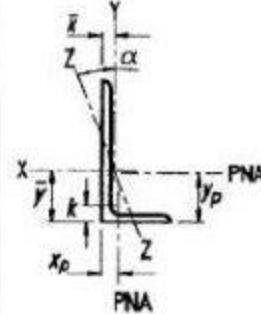
**Effective Net Area  $A_e = UA_n$**

where  $U$  is Shear Lag Factor (a reduction co-efficient).

**$U = 1.0 - \bar{x}/L$  [except plates and HSS sections, AISC Table D3.1]**

**$\bar{x}$  = distance between loading line (c.g. of tension member section) and connection plane.**

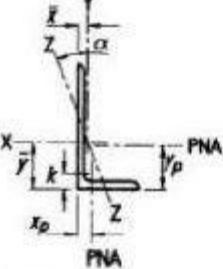
**$L$  = Length of connection**



**Table 1-7  
Angles  
Properties**

Shape	<i>k</i>	Wt.	Area, <i>A</i>	Axis X-X					Flexural-Torsional Properties			
				<i>I</i>	<i>S</i>	<i>r</i>	$\bar{y}$	<i>Z</i>	$y_p$	<i>J</i>	<i>C_w</i>	$\bar{r}_o$
	in.	lb/ft	in. <sup>2</sup>	in. <sup>4</sup>	in. <sup>3</sup>	in.	in.	in. <sup>3</sup>	in.	in. <sup>4</sup>	in. <sup>6</sup>	in.
L6×6×1	1½	37.4	11.0	35.4	8.55	1.79	1.86	15.4	0.918	3.68	9.24	3.18
×7/8	1¾	33.1	9.75	31.9	7.61	1.81	1.81	13.7	0.813	2.51	6.41	3.21
×3/4	1¼	28.7	8.46	28.1	6.64	1.82	1.77	11.9	0.705	1.61	4.17	3.24
×5/8	1⅓	24.2	7.13	24.1	5.64	1.84	1.72	10.1	0.594	0.955	2.50	3.28
×9/16	1⅕	21.9	6.45	22.0	5.12	1.85	1.70	9.18	0.538	0.704	1.85	3.29
×1/2	1	19.6	5.77	19.9	4.59	1.86	1.67	8.22	0.481	0.501	1.32	3.31
×7/16	15/16	17.2	5.08	17.6	4.06	1.86	1.65	7.25	0.423	0.340	0.899	3.32
×3/8	7/8	14.9	4.38	15.4	3.51	1.87	1.62	6.27	0.365	0.218	0.575	3.34
×5/16	13/16	12.4	3.67	13.0	2.95	1.88	1.60	5.26	0.306	0.129	0.338	3.35

# Weight calculations



**Table 1-7 (continued)**  
**Angles**  
**Properties**

Shape	<i>k</i>	Wt.	Area, <i>A</i>	Axis X-X						Flexural-Torsional Properties		
				<i>I</i>	<i>S</i>	<i>r</i>	$\bar{y}$	<i>Z</i>	<i>y<sub>p</sub></i>	<i>J</i>	<i>C<sub>w</sub></i>	$\bar{r}_e$
	in.	lb/ft	in. <sup>2</sup>	in. <sup>4</sup>	in. <sup>3</sup>	in.	in.	in. <sup>3</sup>	in.	in. <sup>4</sup>	in. <sup>5</sup>	in.
L4x3½x½	7/8	11.9	3.50	5.30	1.92	1.23	1.24	3.46	0.497	0.301	0.302	2.03
× <sup>3</sup> / <sub>8</sub>	3/4	9.10	2.67	4.15	1.48	1.25	1.20	2.66	0.433	0.132	0.134	2.06
× <sup>5</sup> / <sub>16</sub>	11/16	7.70	2.25	3.53	1.25	1.25	1.17	2.24	0.401	0.0782	0.0798	2.08
× <sup>1</sup> / <sub>4</sub>	5/8	6.20	1.81	2.89	1.01	1.26	1.14	1.81	0.368	0.0412	0.0419	2.09
L4x3x <sup>5</sup> / <sub>8</sub>	1	13.6	3.89	6.01	2.28	1.23	1.37	4.08	0.810	0.529	0.472	1.91
× <sup>1</sup> / <sub>2</sub>	7/8	11.1	3.25	5.02	1.87	1.24	1.32	3.36	0.747	0.281	0.255	1.94
× <sup>3</sup> / <sub>8</sub>	3/4	8.50	2.48	3.94	1.44	1.26	1.27	2.60	0.683	0.123	0.114	1.97
× <sup>5</sup> / <sub>16</sub>	11/16	7.20	2.09	3.36	1.22	1.27	1.25	2.19	0.651	0.0731	0.0676	1.98
× <sup>1</sup> / <sub>4</sub>	5/8	5.80	1.69	2.75	0.988	1.27	1.22	1.77	0.618	0.0386	0.0356	1.99
L3½x3½x <sup>1</sup> / <sub>2</sub>	7/8	11.1	3.25	3.63	1.48	1.05	1.05	2.66	0.466	0.281	0.238	1.87
× <sup>7</sup> / <sub>16</sub>	13/16	9.80	2.87	3.25	1.32	1.06	1.03	2.36	0.412	0.192	0.164	1.89
× <sup>3</sup> / <sub>8</sub>	3/4	8.50	2.48	2.86	1.15	1.07	1.00	2.06	0.357	0.123	0.106	1.90
× <sup>5</sup> / <sub>16</sub>	11/16	7.20	2.09	2.44	0.969	1.08	0.979	1.74	0.301	0.0731	0.0634	1.92
× <sup>1</sup> / <sub>4</sub>	5/8	5.80	1.69	2.00	0.787	1.09	0.954	1.41	0.243	0.0386	0.0334	1.93

### Tension member design problem:

Select an unequal-leg angle tension member 15 ft long to resist a service dead load of 35 kips and a service live load of 70 kips. Use A36 steel ( $F_y = 36$  ksi,  $F_u = 58$  ksi). The connection is shown in figure below. It shall be connected to a gusset plate using 8 nos.  $\frac{3}{4}$ -in dia. bolts in two rows as shown. Neglect block shear failure mode and follow LRFD principle.

**Solution:**

The factored load is

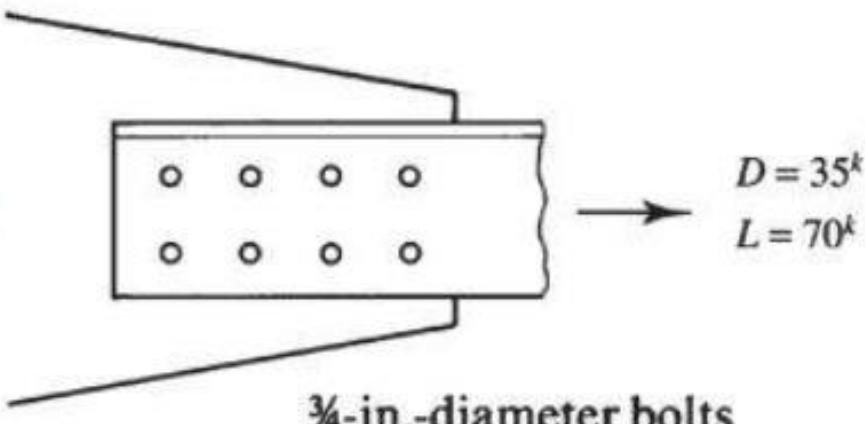
$$\begin{aligned}P_u &= 1.2D + 1.6L = 1.2(35) + 1.6(70) \\&= 154 \text{ kips}\end{aligned}$$

On gross area:  $P_u \leq \phi P_n = \phi F_y A_g$

$$\therefore A_g = P_u / (\phi F_y) = 154 / (0.9 \times 36) = 4.75 \text{ in}^2.$$

On net area:  $P_u \leq \phi P_n = \phi F_u A_e$

$$\therefore A_e = P_u / (\phi F_u) = 154 / (0.75 \times 58) = 3.54 \text{ in}^2.$$



To find the net area  $A_n$ , we need  $U$ . However, at this stage, the section is unknown. We can assume  $U = 0.8$  (Case 7, AISC Table D3.1, four bolts in a row).

$$\text{Thus, } A_n = A_e/U = 3.54/.8 = 4.425 \text{ in}^2.$$

$$\therefore \text{Gross area } A_g = A_n + \text{bolt holes} = 4.425 + 2(3/4 + 1/8)t$$

Where  $t$  is the thickness of the angle which is unknown.

Conservatively, we choose  $t = 3/4$  in

$$\therefore A_g = 4.425 + 2(3/4 + 1/8)(3/4) = 5.74 \text{ in}^2 \text{ (governs)}$$

Now we choose a section from AISC Manual with  $A_g \geq 5.74 \text{ in}^2$  and  $t \leq 3/4$  in. We choose L6 x 4 x 5/8

$$A_g = 5.86 \text{ in}^2, t = 5/8 \text{ in. } \therefore A_n = 5.86 - 2(3/4+1/8)5/8 = 4.77 \text{ in}^2.$$

$$A_e = U A_n = 0.8(4.77) = 3.82 \text{ in}^2.$$

$\therefore$  Yield on gross area,  $\phi P_n = \phi F_y A_g = 0.9(36)5.86 = 189.9$  kip,  
Fracture on effective area,  $\phi P_n = \phi F_u A_e = 0.75(58)3.82 = 166.2$  kip  
Thus, tension capacity  $\phi P_n = 166.2 > P_u$  OK.

How it will be if 3 row bolts!!!

Designation				A	$I_x$	$r_x$	y	$I_y$	$r_y$	x
L 9x4x1	9 x	4 x	1	12.0	97.0	2.84	3.50	12.0	1.00	1.00
L 7x4x0.625	7 x	4 x	5/8	6.48	32.4	2.24	2.46	7.84	1.10	0.963
L 6x4x0.75	6 x	4 x	3/4	6.94	24.5	1.88	2.08	8.68	1.12	1.08
L 6x4x0.625	6 x	4 x	5/8	5.86	21.1	1.90	2.03	7.52	1.13	1.03
L 6x4x0.5625	6 x	4 x	9/16	5.31	19.3	1.90	2.01	6.91	1.14	1.01
L 6x4x0.5	6 x	4 x	1/2	4.75	17.4	1.91	1.99	6.27	1.15	0.987
L 6x4x0.4375	6 x	4 x	7/16	4.18	15.5	1.92	1.96	5.60	1.16	0.964
L 6x4x0.375	6 x	4 x	3/8	3.61	13.5	1.93	1.94	4.90	1.17	0.941
L 4x3.5x0.437	4 x	3 1/2 x	7/16	3.09	4.76	1.24	1.23	3.40	1.05	0.978
L 4x3.5x0.375	4 x	3 1/2 x	3/8	2.67	4.18	1.25	1.21	2.95	1.06	0.955
L 4x3.5x0.312	4 x	3 1/2 x	5/16	2.25	3.56	1.26	1.18	2.55	1.07	0.932
L 4x3.5x0.25	4 x	3 1/2 x	1/4	1.81	2.91	1.27	1.16	2.09	1.07	0.909
L 4x3x0.625	4 x	3 x	5/8	3.98	6.03	1.23	1.37	2.87	0.849	0.871
L 4x3x0.5	4 x	3 x	1/2	3.25	5.05	1.25	1.33	2.42	0.864	0.827
L 4x3x0.4375	4 x	3 x	7/16	2.87	4.52	1.25	1.30	2.18	0.871	0.804
L 4x3x0.375	4 x	3 x	3/8	2.48	3.96	1.26	1.28	1.92	0.879	0.782
L 3.5x2.5x0.25	3 1/2 x	2 1/2 x	1/4	1.44	1.8	1.12	1.11	0.777	0.735	0.614
L 3x2.5x0.5	3 x	2 1/2 x	1/2	2.50	2.08	0.913	1.00	1.30	0.722	0.750
L 3x2.5x0.437	3 x	2 1/2 x	7/16	2.21	1.88	0.920	0.978	1.18	0.729	0.728
L 3x2.5x0.375	3 x	2 1/2 x	3/8	1.92	1.66	0.928	0.956	1.04	0.736	0.706
L 3x2.5x0.312	3 x	2 1/2 x	5/16	1.62	1.42	0.937	0.933	0.898	0.744	0.683
L 3x2.5x0.25	3 x	2 1/2 x	1/4	1.31	1.17	0.945	0.911	0.743	0.753	0.661